

SOCATA
Direction des Services
65921 TARBES CEDEX 9
FRANCE

TELEPHONE: 33 (0)5 62.41.73.00 TELEFAX: 33 (0)5 62.41.76.54

**TB** AIRCRAFT

SB 10-152

55 ATA No

## RECOMMENDED

The technical content of this document is approved under the authority of DOA No. EASA.21J.013

REFERENCE: MODIFICATION No. MOD10-0230-55

**SERVICE BULLETIN** 

**SUBJECT**: HORIZONTAL STABILIZER INTERNAL STRUCTURE

EFFECTIVITY: All TB aircraft

REASON :

A. PROBLEM

Field reports of corrosion on horizontal stabilizer spar.

B. REASON

Humidity accumulation on aircraft subject to severe environmental conditions.

C. SOLUTION

Implementation of a regular inspection of the area.

SUMMARY

A. CREATION OF INSPECTION HOLES

**B. INSPECTION** 

**COMPLIANCE:** 

First application: during the next scheduled inspection.

Next applications (Paragraph B. only): this inspection must be repeated every 6 years.

#### **PROCURABLE MATERIAL:**

None

## **CONSUMABLE MATERIAL** (Local purchase):

- Alodine 1200 (MIL-C-5541) or wash primer
- Epoxy Primer
- Finish paint
- 2 grommets DKK 8/14/18-3 or equivalent
- Clean cloths
- Methyl Ethyl Ketone (M.E.K)
- Anti corrosion Wadis 24 (MIL-C-16173D) or equivalent

MAY 13 Page 1 / 4

**TB** AIRCRAFT

## SERVICE BULLETIN

SB 10-152 55 ATA No.

## **RECOMMENDED**

The technical content of this document is approved under the authority of DOA No. EASA.21J.013

#### TOOLING

- Standard aeronautical maintenance station tools
- Drill, dia.1/8" (3.2 mm)
- Counterbore dia. suitable with endoscope dimensions
- Pneumatic vacuum cleaner
- Endoscope

#### MANPOWER:

1 aeronautical mechanic : 2 hours

#### **TECHNICAL INCIDENCES:**

None

#### **ACCOMPLISHMENT INSTRUCTIONS:**

#### NOTE:

Operations required in this Service Bulletin must be accomplished by persons authorized by their Airworthiness Authorities and according to the procedure described hereafter.

- A. CREATION OF INSPECTION HOLES see Figure 1
  - 1) On the bottom surface, drill the 2 inspection holes (1) to dia. 1/8" (3.2 mm)
  - 2) Bore to a diameter suitable with the endoscope dimensions, the 2 inspection holes (1) and the 2 drainage holes (2) using a counterbore with its associated guide pin 0.62 in. max (16 mm max). Deburr.
  - 3) Vacuum chips and clean area with a clean cloth moistened with M.E.K.
  - 4) Protect bare metal with Alodine 1200 or wash-primer, primer and finish paint.
- B. INSPECTION see Figure 1
  - 1) Insert he endoscope through the 4 holes to check corrosion on the spar.
  - 2) If the inspection does not reveal corroded area, spray Wadis 24 on the spar to protect it.
  - 3) If the inspection reveals a corroded area, contact DAHER SOCATA.

DAHER SOCATA – Direction des Services Aéroport Tarbes – Lourdes – Pyrénées 65921 TARBES Cedex 9 FRANCE

Phone: +33 (0)5 62 41 76 59 Fax: +33 (0)5 62 41 76 54

e-mail: support@socata.daher.com

MAY 13 Page 2 / 4

**TB** AIRCRAFT

# SERVICE BULLETIN

SB 10-152 55

## **RECOMMENDED**

The technical content of this document is approved under the authority of DOA No. EASA.21J.013

## or for the Americas:

SOCATA NORTH AMERICA North Perry Airport 7501 South Airport Rd. Pembroke Pines, FL 33023 USA

Phone: +1 (954) 893 1418 ou 1419

Fax: +1 (954) 964 7092

#### **CAUTION**

## DO NOT PLUG THE DRAINAGE HOLES

- 4) Plug the 2 inspection holes with grommets.
- Make sure all the tools and materials are removed and the work area is clean and free from debris.

#### **UPDATING OF THE AIRCRAFT DOCUMENTATION:**

Upon completion of Service Bulletin No. SB 10-152-55 (Reference MOD10-0230-55) "HORIZONTAL STABILIZER INTERNAL STRUCTURE", make an appropriate maintenance record entry.

WARNING :

SOCATA considers that it is VERY IMPORTANT for operators to comply with the instructions of this SB.

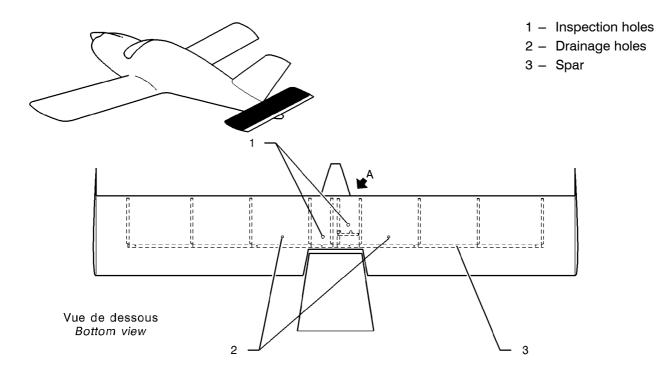
Operators who arbitrarily ignore the compliance statement indicated in this SB do so at their own risk.

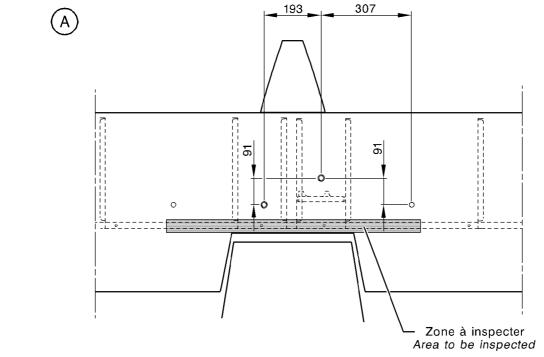
MAY 13 Page 3 / 4

# **SERVICE BULLETIN**

## **RECOMMENDED**

The technical content of this document is approved under the authority of DOA No. EASA.21J.013





14552001AAADYZ4600

Figure 1 - Inspection of horizontal stabilizer internal structure

MAY 13 Page 4 / 4